

Synthesis of Optimal H_∞ Controllers via H_2 -Based Loop-Shaping Design

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Synthesis of optimal H_∞ controllers is formulated as a loop-shaping problem where the desired closed-loop shape to be pursued is a uniform frequency response of the largest singular value. The weighted H_2 optimization technique used in the linear quadratic Gaussian design with loop transfer recovery is exploited in the loop-shaping procedures to generate a sequence of H_2 controllers converging to the optimal H_∞ controller. The resulting optimal H_∞ controller not only has the inherent robust property due to H_∞ criterion but also possesses the nice H_2 control structure, being easy to compute and implement. A fighter example and a large space structure example are demonstrated to show that the numerical accuracy of the present H_2 -based H_∞ synthesis is comparable to the conventional H_∞ approach, i.e., γ -iteration, but with reduced computational efforts.

Nomenclature

$\ A(s)\ _2$	$= [(1/2\pi) \int_{-\infty}^{\infty} \text{tr}[A^*(j\omega)A(j\omega)] d\omega]^{1/2}$
	$= [(1/2\pi) \int_{-\infty}^{\infty} \sum_i \sigma_i^2[A(j\omega)] d\omega]^{1/2}$
$\ A(s)\ _\infty$	$= \sup_{-\infty < \omega < \infty} \bar{\sigma}[A(j\omega)]$
G	= nominal plant
K_i	= optimal H_2 controller at the i th iteration
M	$= F_l(P, K) = P_{11} + P_{12}K(I - P_{22}K)^{-1}P_{21}$
P	= augmented plant, $\begin{bmatrix} P_{11} & P_{12} \\ P_{21} & P_{22} \end{bmatrix}$
R	= control sensitivity function, $K(I + GK)^{-1}$
S	= output sensitivity function, $(I + GK)^{-1}$
T	= complementary sensitivity function, $GK(I + GK)^{-1}$
$W_i(s)$	= scalar weighting function at the i th iteration
W_R, W_S, W_T	= weighting functions associated with R, S, T , respectively
$\gamma_i(\omega)$	$= \bar{\sigma}[W_i F_l(P, K_i)(j\omega)]$
$\zeta_i(\omega)$	$= \bar{\sigma}[F_l(P, K_i)(j\omega)]$
λ_i	$= \ W_i F_l(P, K_i)\ _2$
$\bar{\sigma}(M)$	= largest singular value of M
$\sigma_i(M)$	= i th largest singular value of M

Introduction

It has been recognized for long that the H_∞ approach to control system is very appropriate for the optimization of stability and disturbance rejection robustness properties, whereas the linear quadratic Gaussian (LQG) type of cost function is, on the other hand, often a more practical criterion for minimizing tracking errors or control signal variations, because of reference input changes. Therefore, to quantitatively demonstrate design tradeoffs, the simultaneous treatment of both H_2 and H_∞ performance criteria becomes indispensable. In conjunction with this consideration, mixed H_2/H_∞ problems, such as LQG control with an H_∞ performance bound^{1,2} and mixed H_2 and H_∞ performance objectives^{3,4} have been proposed and solved in the literature.

The purpose of this paper is not to solve the mixed H_2/H_∞ control problems; instead, we will show that a LQG controller is itself an optimal H_∞ controller. The main idea we want to point out here is that arbitrary optimal H_∞ control problems for linear time invariant (LTI) systems can be solved by LQG controllers with proper selection of frequency-dependent weights. Consider the standard

feedback framework for multivariable LTI systems in Fig. 1, where $P(s)$ is the given augmented plant and $K(s)$ is the controller to be designed. The synthesis of optimal H_∞ controllers is to solve the following optimization problem:

$$\inf_{K \text{ stabilizing}} \|F_l(P, K)\|_\infty = \inf_{K \text{ stabilizing}} \sup_{\omega} \bar{\sigma}[F_l(P, K)(j\omega)] \quad (1)$$

where the lower linear fractional transformation (LFT) $F_l(P, K)$ is the transfer function matrix from the exogenous input w to the penalized output z . Let K_0 be the corresponding optimal H_∞ controller; then it is well known that K_0 possesses the following all-pass property:

$$\bar{\sigma}[F_l(P, K_0)(j\omega)] = \text{const} \quad \forall \omega \quad (2)$$

This all-pass property is very useful in characterizing optimal H_∞ solutions. There are many other optimization problems possessing this all-pass property.⁵ In this paper, we introduce loop-shaping design to construct all-pass functions based on the H_2 -optimization technique. The significance of this approach is that the resulting optimal H_∞ controllers can be assigned a priori with LQG structure.

The first attempt to obtain optimal H_∞ controllers by using optimization criterion other than H_∞ norm was made by Kwakernaak.⁶ The key point of his result is because of the following observation. Suppose that there exists a nonnegative, rational, strictly proper matrix $W(j\omega)$, such that when the cost function

$$\int_{-\infty}^{\infty} \text{tr}[W(j\omega)F_l(P, K)(j\omega)] d\omega$$

is minimized by the controller K_0 with $F_l(P, K_0)(j\omega)$ being all pass, then K_0 also minimizes $\|F_l(P, K)\|_\infty$. As shown in the next section, this result can be generalized to the form that there exists a frequency-dependent weight W such that

$$\arg \inf_K \|W F_l(P, K)\|_2 = \arg \inf_K \|F_l(P, K)\|_\infty \quad (3)$$

The left-hand side of Eq. (3) is known as the frequency-weighted LQG problem,⁷⁻⁹ which is intimately related to the LQG/loop

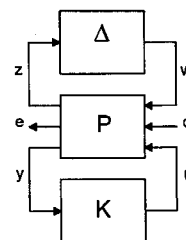


Fig. 1 General framework of feedback control system.

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transfer recovery (LTR) technique.¹⁰ Equation (3) reveals the equivalence between the H_∞ -optimization problem and the frequency-weighted H_2 -optimization problem. For mixed sensitivity problems, Kwakernaak^{6,11} and Grimble¹² derived the frequency-dependent weight $W(s)$ and obtained the corresponding optimal H_∞ controller $K_0(s)$ in Eq. (3). Here we will go one step further by showing in the next section that for any control problems formulated by the LFT $F_l(P, K)$, there always exists a frequency-dependent weight $W(s)$ such that the equivalence in Eq. (3) is achieved.

The optimal H_∞ controller K_0 obtained from the weighted H_2 -optimization process not only has the inherent robust property because of the H_∞ criterion but also possesses the nice H_2 control structure, being easy to compute and implement. The motivation of this approach is to recall the role played by weighted H_2 in the scenario of LQG synthesis with LTR (LQG/LTR) in which weighting functions are adjusted such that the prescribed shape of the open-loop response are met. McFarlane and Glover¹³ proposed a loop-shaping design procedure using H_∞ optimization, where in contrast to open-loop shaping, closed-loop objectives are specified in terms of the requirements on the singular values of the closed-loop transfer functions. In conjunction with these observations, we can interpret the problem raised by Eq. (3) in a loop-shaping manner: given the all-pass requirement (2) on the largest singular value of $F_l(P, K)(j\omega)$, find a loop-shaping design procedure using H_2 optimization to meet this requirement. In other words, we want to shape the H_2 norm of the weighted closed-loop transfer function $W F_l(P, K)$ by adjusting the weight W such that the frequency response of the largest singular value of $F_l(P, K)$ becomes uniform.

H_2 -Based Loop-Shaping Design

Consider the feedback control system shown in Fig. 2, where $w = [d \ n \ r]^T$ is the exogenous input consisting of process disturbance d , measurement noise n , and reference command r ; also, $z = [z_1 \ z_2 \ z_3]^T$ is the penalized output comprising weighted tracking error z_1 , weighted control signal z_2 , and weighted closed-loop response z_3 ; and y is measurement and u is control signal. The input-output relation can be written as

$$\begin{bmatrix} z \\ y \end{bmatrix} = P \begin{bmatrix} w \\ u \end{bmatrix} \quad (4)$$

where P is the augmented plant given by

$$P(s) = \begin{bmatrix} P_{11}(s) & P_{12}(s) \\ P_{21}(s) & P_{22}(s) \end{bmatrix} = \begin{bmatrix} -W_S G & -W_S & W_S & -W_S G \\ 0 & 0 & 0 & W_R \\ W_T G & 0 & 0 & W_T G \\ -G & -I & I & -G \end{bmatrix}$$

The transfer function from the exogenous input w to the penalized output z is found by combining Eq. (4) with the control law $u = K(s)y$, leading to $z = F_l(P, K)w = [P_{11} + P_{12}K(I - P_{22}K)^{-1}P_{21}]w$. The optimal H_∞ control problem is to find controller K to minimize the following cost function:

$$\sup_{\|w\|_2 \leq 1} \|z\|_2 = \|F_l(P, K)\|_\infty \quad (5)$$

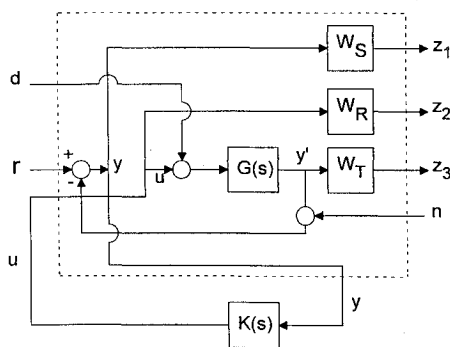


Fig. 2 Block diagram of mixed sensitivity problem.

Here we focus on constructing the optimal H_∞ controller from the approach of weighted H_2 optimization in Eq. (3). The determination of the frequency-dependent weight W in Eq. (3) depends on the structure of $F_l(P, K)$. Two special cases have been considered in the literature. Kwakernaak¹¹ and Grimble¹² obtained W for the mixed sensitivity problem

$$F_l(P, K) = \begin{bmatrix} W_S S \\ W_R R \end{bmatrix} \quad (6a)$$

where $S(s) = [I + G(s)K(s)]^{-1}$ is the output sensitivity matrix and $R(s) = K(s)[I + G(s)K(s)]^{-1}$ is the control sensitivity matrix. The problem considered by Kwakernaak⁶ is for the case

$$F_l(P, K) = \begin{bmatrix} W_S S \\ W_T T \end{bmatrix} \quad (6b)$$

where $T(s) = G(s)K(s)[I + G(s)K(s)]^{-1}$ is the complementary sensitivity function. Their approach first determined the frequency-dependent weight W according to the all-pass condition (2), and then obtained the optimal H_∞ controller by solving an auxiliary LQG problem with dynamic weighting matrices, i.e., by solving an auxiliary weighted H_2 -optimization problem. With their approach, the formulation strongly depends on the structure of $F_l(P, K)$, and the general procedure of finding the desired H_2 frequency-dependent weight W for an arbitrary $F_l(P, K)$ is still lacking in the literature.

In this section we propose a loop-shaping design procedure to construct W and the corresponding optimal H_∞ controller for arbitrary structure of $F_l(P, K)$. The ultimate loop shape we want to achieve is a uniform frequency response of the largest singular value of $F_l(P, K)$, i.e., the loop shape represented by the all-pass condition (2). The strategy we will adopt is to shape W recursively by a sequence of H_2 optimizations until a uniform shape of $\bar{\sigma}[F_l(P, K)(j\omega)]$ is achieved. Notice that although we focus on the uniformity of $\bar{\sigma}[F_l(P, K)(j\omega)]$, the shape of $\bar{\sigma}[S(j\omega)]$ and $\bar{\sigma}[T(j\omega)]$ can be assigned via proper choice of the weighting function $W(s)$. Taking mixed sensitivity problem (6b) for instance, when $\bar{\sigma}[F_l(P, K)(j\omega)]$ becomes uniform, the magnitudes of $\bar{\sigma}[S(j\omega)]$ and $\bar{\sigma}[T(j\omega)]$ are approximately proportional to the inverse magnitudes of $\bar{\sigma}[W_S(j\omega)]$ and $\bar{\sigma}[W_T(j\omega)]$, respectively. Hence, through proper choices of W_S and W_T we can impose the desired constraints on the singular values of S and T .

The underlying concept of the H_2 -based loop shaping design is easy to comprehend. For a scalar frequency-dependent weight $W(s)$, we have the following identity:

$$\bar{\sigma}[W F_l(P, K)(j\omega)] = |W(j\omega)| \bar{\sigma}[F_l(P, K)(j\omega)] \quad (7)$$

It can be rewritten in an alternative way as

$$|W(j\omega)| = \frac{\bar{\sigma}[W F_l(P, K)(j\omega)]}{\bar{\sigma}[F_l(P, K)(j\omega)]} \quad (8)$$

where K is an optimal H_2 controller. Since both $W(s)$ and $K(s)$ are unknown, we can neither determine $W(s)$ from Eq. (8), nor can we determine $K(s)$ from the optimization of $\|W F_l(P, K)\|_2$; however, if we apply Eqs. (7) and (8) to different iteration steps, $K(s)$ and $W(s)$ can be determined iteratively from each other. For example, if we have a sequence of optimal H_2 controllers $(K_0, K_1, \dots, K_i, \dots)$ where we suppose $K_0(s)$ to $K_{i-1}(s)$ and the corresponding frequency-dependent weights $W_0(s)$ to $W_{i-1}(s)$ are known. The objective is to determine $K_i(s)$ and $W_i(s)$. We can exploit $K_{i-1}(s)$ and $W_{i-1}(s)$ to determine $W_i(s)$ by applying Eq. (8):

$$|W_i(j\omega)| = \frac{\bar{\sigma}[W_{i-1} F_l(P, K_{i-1})(j\omega)]}{\|F_l(P, K_{i-1})\|_\infty} \quad (9)$$

where we have replaced $\bar{\sigma}[F_l(P, K_{i-1})(j\omega)]$ in Eq. (8) by $\|F_l(P, K_{i-1})\|_\infty$, since if $K_{i-1}(s)$ is close to the optimal H_∞ controller, we must have $\bar{\sigma}[F_l(P, K_{i-1})(j\omega)] = \|F_l(P, K_{i-1})\|_\infty, \forall \omega$. Next, we use this $W_i(s)$ to determine the optimal H_2 controller

$$K_i = \arg \inf_{K \text{ stabilizing}} \|W_i F_l(P, K)\|_2 \quad (10)$$

As the iteration proceeds, the following sequences can be defined:

$$\lambda_i = \|W_i F_i(P, K_i)\|_2 \quad (11)$$

$$\zeta_i(\omega) = \bar{\sigma}[F_i(P, K_i)(j\omega)] \quad (12)$$

$$\gamma_i(\omega) = \bar{\sigma}[W_i F_i(P, K_i)(j\omega)] \quad (13)$$

where $W_i(s)$ is a scalar, minimum-phase transfer function whose magnitude is determined by $\gamma_{i-1}(\omega)/\|\zeta_{i-1}(\omega)\|_\infty$, i.e.,

$$|W_i(j\omega)| = \frac{\gamma_{i-1}(\omega)}{\|\zeta_{i-1}(\omega)\|_\infty}, \quad i = 1, 2, \dots \quad (14)$$

with $|W_0(j\omega)| = 1$. The sequence of the optimal H_2 controllers converging to an optimal H_∞ controller can be characterized in the following way.

Theorem 1: Given the controller sequence $(K_i)_{i=0}^\infty$ defined by Eq. (10) with W_i given by Eq. (14), then the corresponding sequences $(\lambda_i)_{i=0}^\infty$, $[\zeta_i(\omega)]_{i=0}^\infty$, $[\gamma_i(\omega)]_{i=0}^\infty$ possess the following properties.

Property 1:

$$(1/\sqrt{n})\lambda_i \leq \|\gamma_i(\omega)\|_2 \leq \lambda_i \quad (15)$$

Property 2:

$$\frac{\zeta_{i+1}(\omega)}{\|\zeta_i(\omega)\|_\infty} \gamma_i(\omega) = \gamma_{i+1}(\omega), \quad \forall \omega \quad (16)$$

Property 3:

$$|W_{i+1}(j\omega)| = \frac{\gamma_i(\omega)}{\|\zeta_i(\omega)\|_\infty} = \prod_{k=0}^i \frac{\zeta_k(\omega)}{\|\zeta_k(\omega)\|_\infty}, \quad \forall \omega \quad (17)$$

where n is the rank of $W_i(j\omega)F_i(P, K_i)(j\omega)$.

Proof:

1) Exploiting the relation

$$\begin{aligned} \frac{1}{n} \sum_{k=1}^n \sigma_k^2[W_i F_i(P, K_i)(j\omega)] &\leq \sigma_1^2[W_i F_i(P, K_i)(j\omega)] \\ &\leq \sum_{k=1}^n \sigma_k^2[W_i F_i(P, K_i)(j\omega)] \end{aligned}$$

and integrating with respect to ω from $-\infty$ to ∞ , we have

$$\begin{aligned} &\sqrt{\frac{1}{n} \int_{-\infty}^{\infty} \sum_{k=1}^n \sigma_k^2[W_i(j\omega)F_i(P, K_i)(j\omega)] d\omega} \\ &\leq \sqrt{\int_{-\infty}^{\infty} \sigma_1^2[W_i(j\omega)F_i(P, K_i)(j\omega)] d\omega} \\ &\leq \sqrt{\int_{-\infty}^{\infty} \sum_{k=1}^n \sigma_k^2[W_i(j\omega)F_i(P, K_i)(j\omega)] d\omega} \end{aligned}$$

From the definition of the H_2 norm, these inequalities lead to the desired result,

$$(1/\sqrt{n})\|W_i F_i(P, K_i)\|_2 \leq \|\sigma_1[W_i F_i(P, K_i)]\|_2 \leq \|W_i F_i(P, K_i)\|_2$$

2) From the definition of $\gamma_{i+1}(\omega)$, we have

$$\begin{aligned} \gamma_{i+1}(\omega) &= \bar{\sigma}[W_{i+1}(j\omega)F_i(P, K_{i+1})(j\omega)] \\ &= |W_{i+1}(j\omega)|\bar{\sigma}[F_i(P, K_{i+1})(j\omega)] \\ &= \frac{\gamma_i(\omega)}{\|\zeta_i(\omega)\|_\infty} \zeta_{i+1}(\omega) \end{aligned}$$

3) By applying Eq. (16) repeatedly, we have

$$\begin{aligned} |W_{i+1}(\omega)| &= \frac{\gamma_i(\omega)}{\|\zeta_i(\omega)\|_\infty} = \frac{\zeta_i(\omega)}{\|\zeta_i(\omega)\|_\infty} \frac{\gamma_{i-1}(\omega)}{\|\zeta_{i-1}(\omega)\|_\infty} \\ &= \frac{\zeta_i(\omega)}{\|\zeta_i(\omega)\|_\infty} \frac{\zeta_{i-1}(\omega)}{\|\zeta_{i-1}(\omega)\|_\infty} \frac{\gamma_{i-2}(\omega)}{\|\zeta_{i-2}(\omega)\|_\infty} \\ &\vdots \\ &= \prod_{k=0}^i \frac{\zeta_k(\omega)}{\|\zeta_k(\omega)\|_\infty} \quad \text{QED} \end{aligned}$$

Now we are ready to show that the limit controller K_∞ in the sequence of the optimal H_2 controllers $(K_i)_{i=0}^\infty$ defined in Eq. (10) is the solution of the H_∞ -optimization problem stated in Eq. (1). The proof contains three steps.

1) The first step is to show the convergence of the sequences $(\lambda_i)_{i=0}^\infty$ and $[\gamma_i(\omega)]_{i=0}^\infty$.

2) The second step is to show that the limit controller K_∞ satisfies the all-pass property: $\bar{\sigma}[F_i(P, K_\infty)(j\omega)] = \zeta_\infty, \forall \omega$.

3) The third step is to show that the limit controller K_∞ truly achieves the infimum:

$$\inf_K \|F_i(P, K)\|_\infty = \|F_i(P, K_\infty)\|_\infty = \zeta_\infty$$

The convergence of the sequences $(\lambda_i)_{i=0}^\infty$ and $[\gamma_i(\omega)]_{i=0}^\infty$ is proved first.

Theorem 2:

1) The sequence $(\lambda_i)_{i=0}^\infty$ is convergent.

2) Let K_i be the central solution¹⁴ of the H_2 -optimization problem defined in Eq. (10); then the controller sequence $(K_i)_{i=0}^\infty$ is convergent.

3) The function sequence $[\gamma_i(\omega)]_{i=0}^\infty$ is convergent.

Proof:

1) By definition [from Eq. (17)],

$$\begin{aligned} \lambda_i &= \left\| \prod_{k=0}^{i-1} \frac{\zeta_k(\omega)}{\|\zeta_k\|_\infty} F_i(P, K_i) \right\|_2 \\ &= \left\| \frac{\zeta_i(\omega)}{\|\zeta_i\|_\infty} \left\| \prod_{k=0}^{i-1} \frac{\zeta_k(\omega)}{\|\zeta_k\|_\infty} F_i(P, K_i) \right\|_2 \right\|_\infty \\ &\quad \left(\text{note } \left\| \frac{\zeta_i(\omega)}{\|\zeta_i\|_\infty} \right\|_\infty = 1 \right) \\ &\geq \left\| \prod_{k=0}^i \frac{\zeta_k(\omega)}{\|\zeta_k\|_\infty} F_i(P, K_i) \right\|_2 \\ &\geq \inf_K \left\| \prod_{k=0}^i \frac{\zeta_k(\omega)}{\|\zeta_k\|_\infty} F_i(P, K) \right\|_2 = \lambda_{i+1} \end{aligned}$$

Hence, the sequence $(\lambda_i)_{i=0}^\infty$ is monotonically decreasing and bounded by $0 \leq \lambda_i \leq \lambda_0$. Employing the convergent theorem¹⁵ of real sequences, we thus obtain the convergence of $(\lambda_i)_{i=0}^\infty$.

2) The solution K_i of the optimal H_2 -optimization problem (10) is, in general, not unique. Nevertheless, if we confine K_i to be the central solution derived in Ref. 14, the one-to-one correspondence between K_i and λ_i can be established. Under such circumstance, the convergence of λ_i implies the convergence of K_i .

3) From the definition, we know that the sequence $[\gamma_i(\omega)]_{i=0}^\infty$ is uniquely determined by the controller sequence $(K_i)_{i=0}^\infty$. Therefore, the convergence of $\gamma_i(\omega)$ is guaranteed by the convergence of K_i .

QED

Having proved the convergence of the two related sequences, we proceed to verify the uniformity of the largest singular value of $F_i(P, K_\infty)(j\omega)$.

Lemma 1: Let K_∞ be the limit of the controller sequence defined in Eq. (10); then

$$\bar{\sigma}[F_l(P, K_\infty)(j\omega)] = \|\zeta_\infty(\omega)\|_\infty = \text{const} \quad \forall \omega \quad (18)$$

Proof: From Eq. (16), we have

$$\lim_{i \rightarrow \infty} \frac{\zeta_i(\omega)}{\|\zeta_{i-1}(\omega)\|_\infty} = \lim_{i \rightarrow \infty} \frac{\gamma_i(\omega)}{\gamma_{i-1}(\omega)}$$

Recall the convergence of $\gamma_i(\omega)$, we obtain $\lim_{i \rightarrow \infty} \gamma_i(\omega)/\gamma_{i-1}(\omega) = 1$, $\forall \omega$. It turns out that $\lim_{i \rightarrow \infty} \zeta_i(\omega)/\|\zeta_{i-1}(\omega)\|_\infty = 1 \forall \omega$, that is, $\zeta_\infty(\omega) = \bar{\sigma}[F_l(P, K_\infty)(j\omega)] = \|\zeta_\infty(\omega)\|_\infty = \text{const}$, $\forall \omega$.

QED

The final step in our characterization of the optimal H_∞ controller via H_2 -based loop-shaping design is to show that K_∞ achieves the infimum: $\inf_K \|F_l(P, K)\|_\infty = \|F_l(P, K_\infty)\|_\infty$. The justification of using the optimal H_2 controllers to synthesize the optimal H_∞ controllers mainly relies on the following theorem.

Theorem 3: Suppose there exists a scalar frequency-dependent weight $W(s)$ such that the optimal H_2 controller K_0 obtained from

$$K_0 = \arg \inf_K \|W F_l(P, K)\|_2 \quad (19)$$

satisfies the all-pass condition: $\bar{\sigma}[F_l(P, K_0)(j\omega)] = \zeta = \text{const}$, $\forall \omega$, then K_0 minimizes the H_∞ -norm criterion: $\inf_K \|F_l(P, K)\|_\infty = \|F_l(P, K_0)\|_\infty = \zeta$.

Proof: We will prove by contradiction. Assume there exists a controller K'_0 satisfying

$$\|F_l(P, K'_0)\|_\infty < \zeta = \bar{\sigma}[F_l(P, K_0)(j\omega)], \quad \forall \omega \quad (20)$$

then we have

$$\begin{aligned} \|W F_l(P, K'_0)\|_2 &\leq \|W\|_2 \|F_l(P, K'_0)\|_\infty \\ &< \zeta \|W\|_2 = \|\zeta W\|_2 \\ &= \|W(j\omega) \bar{\sigma}[F_l(P, K_0)(j\omega)]\|_2 \\ &= \|\bar{\sigma}[W F_l(P, K_0)(j\omega)]\|_2 \end{aligned} \quad (21)$$

By the definition of the H_2 norm,

$$\begin{aligned} \|\bar{\sigma}[W F_l(P, K_0)(j\omega)]\|_2 &= \sqrt{\frac{1}{2\pi} \int_{-\infty}^{\infty} \sigma_1^2[W F_l(P, K_0)(j\omega)] d\omega} \\ \|W F_l(P, K_0)\|_2 &= \sqrt{\frac{1}{2\pi} \int_{-\infty}^{\infty} \sum_{k=1}^n \sigma_k^2[W F_l(P, K_0)(j\omega)] d\omega} \end{aligned}$$

It follows that $\|\bar{\sigma}[W F_l(P, K_0)(j\omega)]\|_2 \leq \|W F_l(P, K_0)\|_2$. Using this result in Eq. (21), we have $\|W F_l(P, K'_0)\|_2 < \|W F_l(P, K_0)\|_2$. But this violates the fact in Eq. (19). Hence, the assumption made in Eq. (20) is incorrect, i.e., the infimum is truly achieved by K_0 .

QED

In Lemma 1 we have shown that the limit controller K_∞ obtained from $K_\infty = \arg \inf_K \|W_\infty F_l(P, K)\|_2$ satisfies the all-pass condition (18). Hence from Theorem 3, we recognize K_∞ as the desired optimal H_∞ controller, and the associated frequency-dependent weight $W(s)$ in Theorem 3 is $W_\infty(s)$, which is the limit of the sequence $(W_i)_{i=0}^\infty$ defined in Eq. (14).

H_2 -Based H_∞ -Synthesis Algorithm

Using Eq. (17), the controller sequence $(K_i)_{i=0}^\infty$ defined in Eq. (10) can be rewritten in the following manner:

$$\begin{aligned} K_i &= \arg \inf_K \|W_i F_l(P, K)(j\omega)\|_2 \\ &= \arg \inf_K \left\| \prod_{k=0}^{i-1} \frac{\zeta_k(\omega)}{\|\zeta_k(\omega)\|_\infty} F_l(P, K)(j\omega) \right\|_2 \\ &= \arg \inf_K \left\| \prod_{k=0}^{i-1} \zeta_k(\omega) F_l(P, K)(j\omega) \right\|_2 \end{aligned} \quad (22)$$

If we define a new scalar minimum-phase function \hat{W}_i as

$$|\hat{W}_i(j\omega)| = \prod_{k=0}^{i-1} \zeta_k(\omega) \quad (23)$$

then Eq. (22) becomes $K_i = \arg \inf_K \|\hat{W}_i F_l(P, K)\|_2$. Instead of the frequency-dependent weight $W_i(s)$ defined in Eq. (14), we can use $\hat{W}_i(s)$ as a new frequency-dependent weight. The following work is to derive the recursive formula for \hat{W}_i :

$$\begin{aligned} |\hat{W}_{i+1}(j\omega)| &= \prod_{k=0}^i \zeta_k(\omega) = \left(\prod_{k=0}^{i-1} \zeta_k(\omega) \right) \zeta_i(\omega) \\ &= |\hat{W}_i(j\omega)| \bar{\sigma}[F_l(P, K_i)(j\omega)] \\ &= \bar{\sigma}[\hat{W}_i F_l(P, K_i)(j\omega)] \end{aligned} \quad (24)$$

In terms of the new frequency-dependent weight $\hat{W}_i(s)$, we summarize the H_∞ -synthesis technique using H_2 -based loop-shaping procedures in the following algorithm.

Initialization: $i = 0$.

1) Set $\hat{W}_0(s) = 1$.

2) Compute $K_0 = \arg \inf_K \|\hat{W}_0 F_l(P, K)\|_2$ using H_2 -optimization technique.

3) Compute $\bar{\sigma}[\hat{W}_0 F_l(P, K_0)(j\omega)]$.

Recursive formula: $i = 1, 2, 3, \dots$

1) Set $\deg(\hat{W}_i) = n_w$, where \hat{W}_i is a scalar, minimum-phase function.

2) Fit $\bar{\sigma}[\hat{W}_{i-1} F_l(P, K_{i-1})(j\omega)]$ by $|\hat{W}_i(j\omega)|$ [from Eq. (24)].

3) Compute $K_i = \arg \inf_K \|\hat{W}_i F_l(P, K)\|_2$.

4) Compute $\bar{\sigma}[\hat{W}_i F_l(P, K_i)(j\omega)]$.

Repeat the algorithm until the required accuracy in the uniformity of $\zeta_i(\omega) = \bar{\sigma}[F_l(P, K_i)(j\omega)]$ is met. Ideally, at the end of iteration $\bar{\sigma}[F_l(P, K_\infty)(j\omega)]$ must be an all-pass function, as shown in Lemma 1. This algorithm is a new scheme for optimal H_∞ synthesis. Although the convergence of the sequence to the optimal H_∞ controller is guaranteed theoretically, in numerical calculation, the achievable degree of optimization depends on the accuracy of calculating the optimal H_2 controller and on the accuracy of curve fitting. The order of the H_2 controllers in the minimizing sequence can be chosen a priori according to the required closeness to the optimal H_∞ solution. Once controller order is assigned, the order n_w of \hat{W}_i in curve fitting can be determined accordingly.

The multiplication of $F_l(P, K_i)$ by a scalar function \hat{W}_i can be considered as a shaping effect on the augmented plant P by noting that $\hat{W}_i F_l(P, K) = F_l(P_i, K)$, where

$$P_i(s) = \begin{bmatrix} \hat{W}_i(s) I_l & 0 \\ 0 & I_m \end{bmatrix} P$$

with l and m being the dimensions of the output vector z and the measurement vector y , respectively. In this way, a scalar weighting function, instead of matrix-valued weighting function, can always be used effectively, regardless of the dimension of the augmented plant.

In summary, the procedures of synthesizing the optimal H_∞ controllers via the H_2 -based loop-shaping design involve only two steps: one step is the refinement of the frequency-dependent weight via $|\hat{W}_i(j\omega)| = \bar{\sigma}[\hat{W}_{i-1} F_l(P, K_{i-1})(j\omega)]$; the other step is the refinement of the optimal H_2 controller via $K_i(s) = \arg \inf_K \|\hat{W}_i F_l(P, K)\|_2$.

Numerical Examples

These H_2 -based loop-shaping procedures are applied to design H_∞ controllers for two real plants, a modern fighter^{9,16} and a large space structure.^{16,17} These two standard benchmark problems, which have been solved in detail by conventional γ -iteration approach, provide the comparison basis between the present method and the γ -iteration technique.

Fighter H_∞ Design Example

This section contains an example of H_∞ synthesis as applied to the pitch-axis controller design of an experimental highly maneuverable airplane HIMAT.⁹ The linearized model of HIMAT consists of six states: $x^T = (\delta v, \alpha, q, \theta, \alpha_1, \alpha_2)$, representing the forward velocity, angle of attack, pitch rate, pitch angle, and actuator states, respectively. The control inputs are elevon δ_e and canard δ_c . The variables to be measured are α and θ . The state-space nominal model for this two-input two-output plant is given by

$$G(s) = \begin{bmatrix} A & B \\ C & D \end{bmatrix} = \begin{bmatrix} -0.023 & -37 & -19 & -32 & 3.3 & -0.76 & 0 & 0 \\ 0 & -1.9 & 0.98 & 0 & -0.17 & 0 & 0 & 0 \\ 0.012 & 12 & -2.6 & 0 & -32 & 22 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & -30 & 0 & 30 & 0 \\ 0 & 0 & 0 & 0 & 0 & -30 & 0 & 30 \\ 0 & 1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 \end{bmatrix}$$

The singular value design specifications are as follows.

1) The robustness specification is -40 dB/decade roll-off and at least -20 dB at 100 rad/s. The weighting function W_S reflecting this specification can be chosen as¹⁶

$$W_S(s) = 16.8 \begin{bmatrix} \frac{s+100}{100s+1} & 0 \\ 0 & \frac{s+100}{100s+1} \end{bmatrix}$$

2) For the performance specification, minimize the sensitivity function as much as possible. The associated weighting function W_T is chosen as¹⁶

$$W_T(s) = \begin{bmatrix} \frac{s^2}{1000} & 0 \\ 0 & \frac{s^2(\tau s + 1)}{1000} \end{bmatrix}$$

where $\tau = 0.5$ ms is selected such that both channels are penalized equally up to $1/\tau$ rad/s. Note that because W_T is an improper transfer function, it cannot be realized in state-space form; but $W_T(s)G(s)$ is proper. This particular $W_T(s)$ ensures that the D_{12} matrix of the augmented plant $P(s)$ is full rank. Augment the nominal plant $G(s)$ with the weighting functions $W_S(s)$ and $W_T(s)$ to form the augmented plant $P(s)$ as

$$P(s) = \begin{bmatrix} W_S & -W_S G \\ 0 & W_T G \\ I & -G \end{bmatrix}$$

and the H_∞ -optimization problem is to find K to minimize $\|F_i(P, K)\|_\infty$. The interconnection of the frequency-dependent weight $\hat{W}_i(s)$ with the original system is demonstrated in Fig. 3. We will shape the weighting functions $W_S(s)$ and $W_T(s)$ such that the H_∞ -optimization problem can be converted to an equivalent weighted H_2 -optimization problem with the reshaped weighting functions given by $\hat{W}_i(s)W_S(s)$ and $\hat{W}_i(s)W_T(s)$. Different order of \hat{W}_i and different number of iterations can be exploited to test the

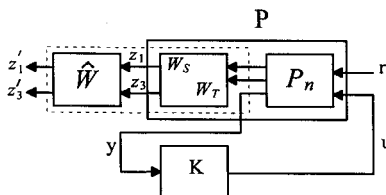


Fig. 3 Block diagram with auxiliary weighting function $\hat{W}(s)$.

Table 1 Fighter example results

No.	$\ \hat{W}_i F_i(P, K_i)\ _2$	$\ F_i(P, K_i)\ _\infty$
0	7.1032	1.4830
1	0.9657	1.0827
2	0.9643	0.9813
3	0.9650	0.9727
4	0.9648	0.9663

accuracy and the converging tendency of the proposed scheme. For example, we use fourth-order $\hat{W}_i(s)$ for curve fitting. Four iterations have been performed and yield five optimal H_2 controllers:

$$K_i = \arg \inf_K \|\hat{W}_i F_i(P, K)\|_2, \quad i = 0, 1, 2, 3, 4 \quad (25)$$

Numerical results for $\|\hat{W}_i F_i(P, K_i)\|_2$ and $\|F_i(P, K_i)\|_\infty$, $i = 0, 1, 2, 3, 4$ are shown in Table 1.

The plot of $\zeta_i(\omega) = \bar{\sigma}[F_i(P, K_i)(j\omega)]$ is shown in Fig. 4, where we can observe that K_4 is very close to the optimal H_∞ controller rendering $\bar{\sigma}[F_i(P, K_4)(j\omega)] = \text{const}, \forall \omega$. The weighting functions W_S and W_T before and after reshaping by \hat{W}_4 are shown in Figs. 5 and 6, respectively. If we fit \hat{W}_i with higher degrees and increase the computational accuracy, the sequence of the optimal H_2 controllers K_i can be made arbitrarily close to the optimal H_∞ solution. This example illustrates, as proved earlier, that by recursive frequency shaping, the optimal H_2 controllers can converge gradually to the optimal H_∞ controller. From the viewpoint of numerical computation, the main advantage of the present H_∞ synthesis technique is its fast convergent speed with reduced computational effort, when compared with the γ -iteration technique. The differences between these two schemes are listed next.

1) To start γ iteration, we need an initial guess for the H_∞ norm. Hence, the required iteration number depends on the closeness between the initial guess and the optimal solution. However, no initial guess is required in the present scheme.

2) The accuracy of the γ -iteration scheme depends on the error tolerance set by the users. The error tolerance stands for the difference of the γ values between the two consecutive iterations. For instance, if we set the error tolerance to 0.01 , the typical required iteration number is about 10, whereas if we reduce the error tolerance to 0.001 , the typical required iteration number is about 20. However, the accuracy of the present scheme does not have as strong a dependence on the iteration number as that of the γ iteration scheme. In the many case studies conducted by the authors, the present scheme appears to achieve its steady state after four or five iterations. Iteration after five is often unnecessary, since the output of the algorithm does not change within the given error tolerance. The closeness of the steady-state value to the optimal H_∞ norm depends on the accuracy of curve fitting. In short, the accuracy of the γ -iteration scheme is dominated by the number of iterations, whereas the accuracy of the present scheme is dominated by the accuracy of curve fitting technique.

3) At each iteration, the γ -iteration scheme solves a suboptimal H_∞ control problem, whereas the present scheme solves an optimal H_2 control problem. These two control problems¹⁴ are governed by two similar pairs of algebraic Riccati equations (ARE). The computational time to solve the H_∞ AREs and the H_2 AREs is quite the same.

For the purpose of the comparison, the example is resolved by the γ -iteration scheme. An initial guess of the optimal H_∞ norm is required to start the γ -iteration process. The value $\lambda_0 = \|F_i(P, K_0)\|_\infty = 1.483$, with K_0 given by Eq. (25), serves as a good starting point for γ iteration. The error tolerance is set to 0.01 , and the γ -iteration process is completed after 11 iterations. The result is shown in Fig. 4, where we can see that the singular value plots of the two schemes are nearly indistinguishable. The H_∞ norms achieved by the γ -iteration scheme and the present scheme are 0.9642 and 0.9663 , respectively. However, although the resulting accuracy is comparable, the γ -iteration scheme achieves this accuracy by solving the H_2 AREs a single time (for initial guess) and the H_∞ AREs 11 times, whereas the present scheme solves the H_2 AREs only 5 times.

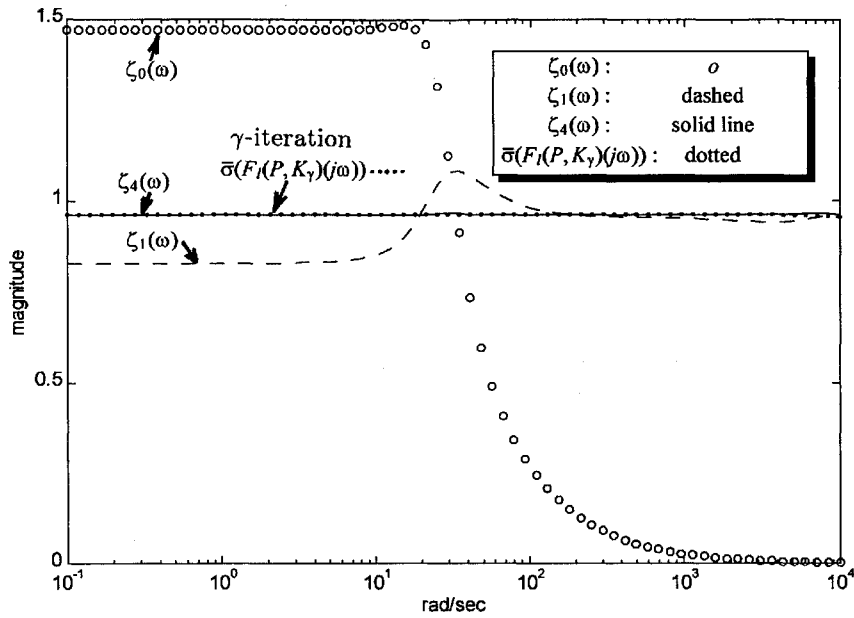


Fig. 4 Iteration results of the fighter design example.

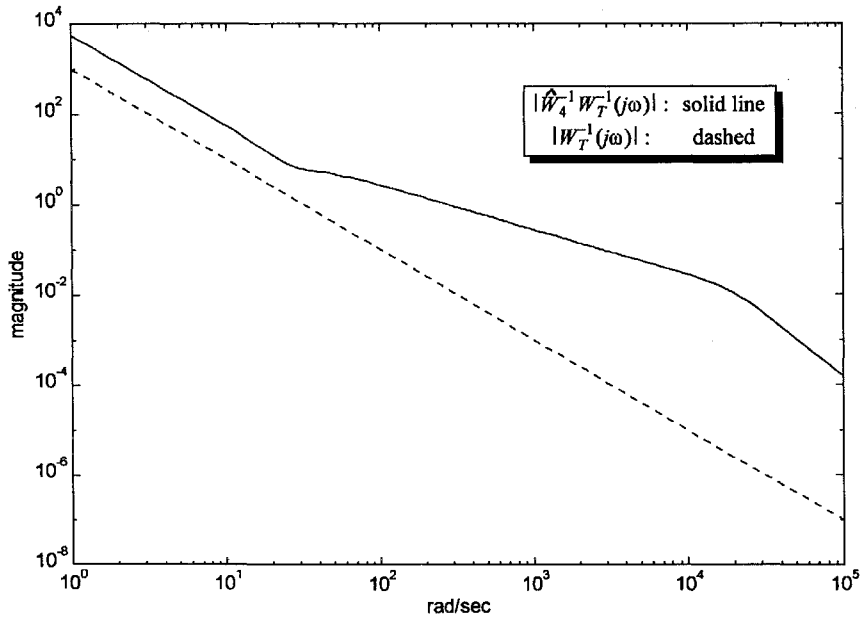


Fig. 5 Reshaping of $W_T^{-1}(s)$ by $\hat{W}_4^{-1}(s)$ of fighter example.

Large Space Structure Design Example

The large space structure (LSS) model¹⁷ was generated by the NASTRAN finite element program by TRW Space Technology Group. It consists of 58 vibrational modes and is controlled by 18 actuators and 20 sensors. To simulate the real environmental vibration source, 12 disturbances are acting on the top and the bottom of the structure. This leads to a state-space representation of the form $\dot{x} = Ax + Bu$, $y = Cx$, where $A \in \mathbf{R}^{116 \times 116}$, $B \in \mathbf{R}^{116 \times 30}$, and $C \in \mathbf{R}^{20 \times 116}$. The four-state approximation of the plant with square-down filter is given by

$$G(s) =$$

$$\begin{bmatrix} -0.9900 & 0.0005 & 0.4899 & 1.9219 & 0.7827 & -0.6140 \\ 0.0009 & -0.9876 & 1.9010 & -0.4918 & 0.6130 & 0.7826 \\ -0.4961 & -1.9005 & -311.70 & 4.9716 & 0.7835 & 0.5960 \\ -1.9215 & 0.4907 & -7.7879 & -398.31 & 0.6069 & -0.7878 \\ 0.7829 & 0.6128 & -0.7816 & -0.6061 & 0 & 0 \\ -0.6144 & 0.7820 & -0.5984 & 0.7884 & 0 & 0 \end{bmatrix}$$

The LSS design specification requires the line-of-sight error to be attenuated at least 100:1 at frequency from 0 to 15 Hz after the feedback control loop is closed. Allowing for a 30 dB per decade roll-off beyond 15 Hz places the control loop bandwidth of roughly 300 Hz. These specifications lead to the following weighting functions.¹⁶

1) The robustness specification is -20 dB/decade roll-off above 2000 rad/s:

$$W_T(s) = \begin{bmatrix} \frac{s}{2000} & 0 \\ 0 & \frac{s}{2000} \end{bmatrix}$$

2) For the performance specification, minimize the sensitivity function

$$W_S(s) = 1.4 \begin{bmatrix} \frac{(1+s/5000)^2}{0.01(1+s/100)^2} & 0 \\ 0 & \frac{(1+s/5000)^2}{0.01(1+s/100)^2} \end{bmatrix}$$

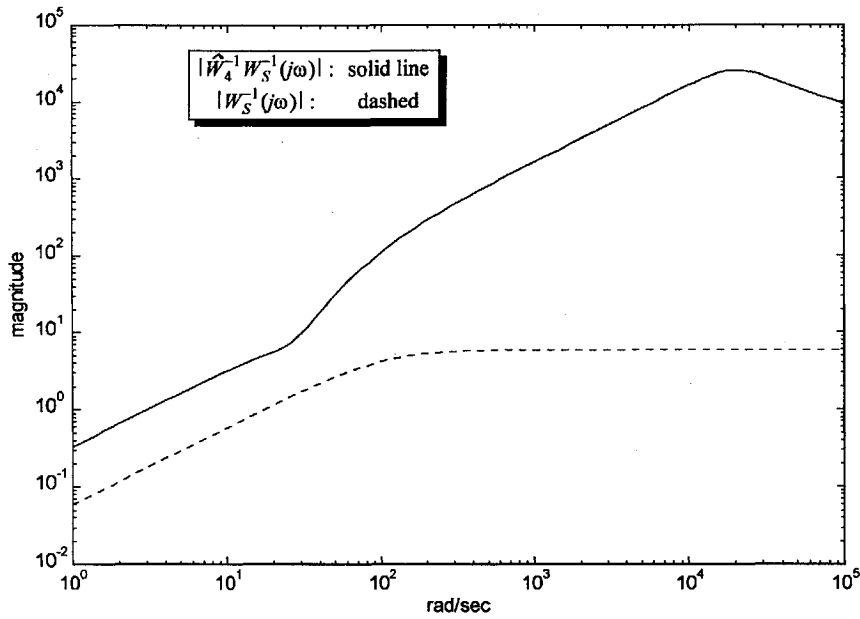


Fig. 6 Reshaping of $W_S^{-1}(s)$ by $\hat{W}_4^{-1}(s)$ of fighter example.

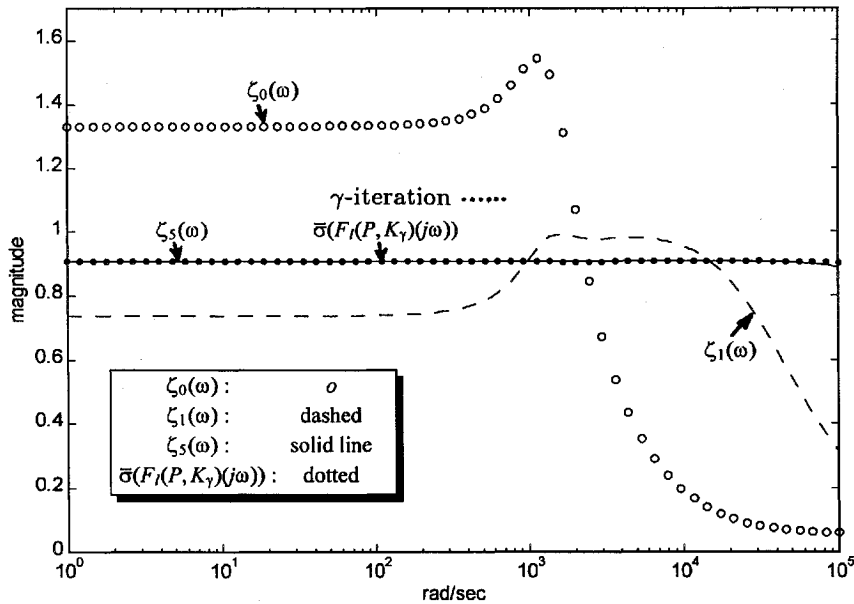


Fig. 7 Iteration results of large space structure design example.

Table 2 LSS example results

No.	$\ \hat{W}_i F_l(P, K_i)\ _2$	$\ F_l(P, K_i)\ _\infty$
0	114.41	1.5445
1	0.4916	0.9915
2	0.9106	0.9317
3	0.9139	0.9176
4	0.8999	0.9089
5	0.9068	0.9074

After augmentation with W_S and W_T , the augmented plant $P(s)$ has eight states. This design problem is reduced to the minimization of $\|F_l(P, K)\|_\infty$, which, in turn, is solved by the H_2 -based loop-shaping procedures proposed previously. After five iterations, we obtained the results given in Table 2.

The frequency responses of $\bar{\sigma}[F_l(P, K_i)(j\omega)]$ are depicted in Fig. 7, where we can see that $\bar{\sigma}[F_l(P, K_s)(j\omega)]$ keeps constant up to the frequency 10^5 rad/s with the constant value (H_∞ norm) given by 0.9074. Also shown in Fig. 7 is the result from the γ -iteration scheme, where the error tolerance is set to 0.01 and the initial guess

of the optimal H_∞ norm is given by $\lambda_0 = \|F_l(P, K_0)\|_\infty = 1.5445$. The γ -iteration process is completed after 12 iterations, and the resulting H_∞ norm is 0.905. As in the previous example, this example reveals that the accuracy of the present scheme by solving the H_2 AREs 6 times is comparable to the accuracy of the γ -iteration scheme by solving the H_2 AREs 1 time and the H_∞ AREs 12 times.

Because the weighted H_2 -optimization problems can be solved by LQG/LTR design procedure,¹⁰ we can also implement the proposed H_2 -based loop-shaping design in terms of the LQG/LTR procedure to obtain the optimal H_∞ controllers.

Conclusions

The H_∞ -optimization problem has been solved using H_2 -based loop-shaping formulation where the desired closed-loop shape to be pursued is a uniform frequency response of the largest singular value. Along this approach, we have verified the possibility that conventional LQG controllers, with appropriate selection of frequency-dependent weights, can become optimal H_∞ controllers. By providing a systematic methodology of determining the frequency-dependent weights via loop-shaping procedures

proposed here, we have derived, both theoretically and numerically, a sequence of H_2 controllers converging to the optimal H_∞ controllers. Without going into the details of the profound H_∞ control theory, the proposed workable algorithm provides an easy access to the optimal H_∞ controllers for engineers who are familiar with the LQG or H_2 control design.

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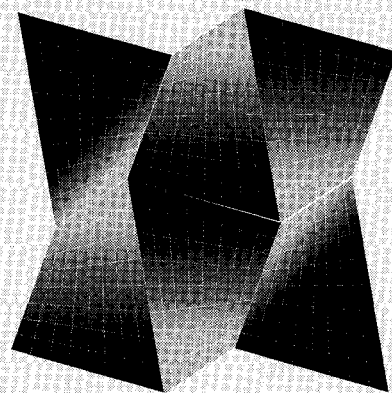
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